

# Multi-concern Dependability-centered Assurance for Space Systems via ConcertoFLA

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Space systems need to be engineered in compliance with ECSS standards and need to ensure a certain degree of dependability. European Cooperation for Space Standardization (ECSS) provides standards for engineering, management and qualification of space systems. More specifically, ECSS has dedicated standards for product assurance for dependability and safety, i.e. ECSS-Q-ST-30C [2] and ECSS-Q-ST-40C [4] respectively. Assurance of all these properties, require the modelling of various system characteristics and their co-analysis, in order to enable the management of trade-offs between them.

CHES [1] an open source supporting toolset, which is the result of several R&D projects, promotes a component-based development methodology in which particular emphasis is given to the ability to specify the non-functional properties of components, including critical properties such as real-time and dependability related characteristics. In particular, ConcertoFLA [2] [5], which is part of CHES toolset, allows users (system architects and dependability engineers) to decorate component-based architectural models with dependability-related information, execute Failure Logic Analysis (FLA) techniques, and get the results back-propagated onto the original model. ConcertoFLA is built on top of Failure Propagation Transform Logic (FPTC) [7] and CHESFLA [10]. ConcertoFLA is a compositional technique to qualitatively assess the dependability of component-based systems, and partially combines and automatizes traditional safety analysis techniques (e.g., Fault Tree Analysis). Its analysis results have also been exploited to semi-automatically generate arguments fragments for safety assurance [8].

In this paper, we present the customization of the CHES methodology [11] and ConcertoFLA in the context of the ECSS standards to enable architects and dependability experts to define a system and perform dependability-centered co-analysis for assuring the required non-functional properties of the system according to ECSS requirements. Figure 1 shows a high-level view of the overall workflow of the proposed customization. The initial step is to define the system by modelling its components and the interactions. Then, the system design is annotated with information specifically related to system reliability, safety and security, associated directly with the involved architectural elements, at the chosen level of architectural detail, and the overall dependability analysis is performed and its results are back propagated. In the next step, the results of the analysis are interpreted for multi-concern e.g., reliability, safety and security. Based on this interpretation a decision is made for introducing the dependability means by refactoring the system. This process is iterated, until the sufficient level of these concerns is met. Further, the paper reports on the usage of the proposed customization in the context of the Attitude Control Systems Engineering (ACS). The ACS plays an important role within satellites, by contributing to maintaining the orientation of the satellite in three-dimensional space and needs to be engineered in compliance with the ECSS standards to ensure a certain degree of dependability. Figure 2 shows a component-based architecture of ACS, where the Sun sensor measurements and angular velocity are provided as an input and the system computes the control torque to be applied on the satellite body to achieve the target attitude.

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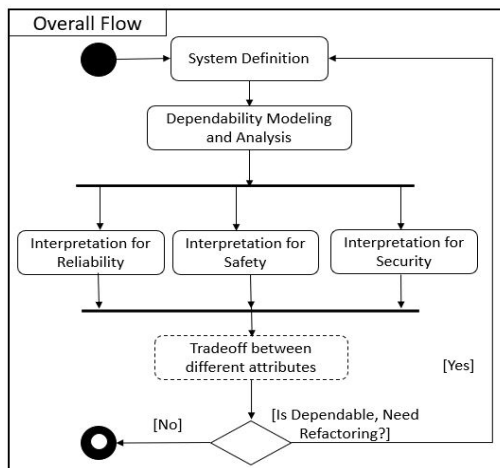


Figure 1 Dependability Co-Analysis via CHES

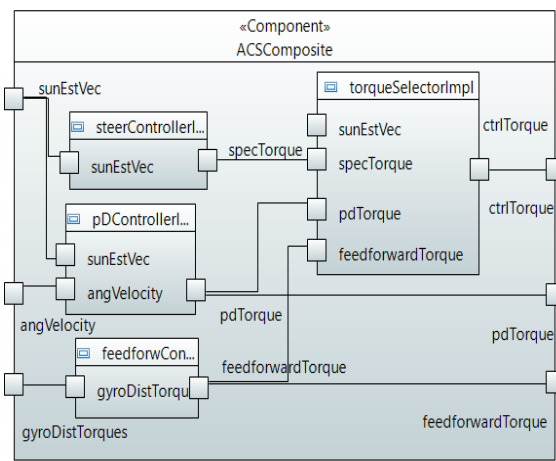


Figure 2 CHES based architecture of ACS

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